

# Document details

< Back to results | 1 of 12 Next >

Export Download Print E-mail Save to PDF Add to List More... >

[View at Publisher](#)

IOP Conference Series: Materials Science and Engineering  
Volume 370, Issue 1, 1 June 2018, Article number 012052  
1st International Conference on Aerospace and Mechanical Engineering, AeroMech 2017;  
Parkroyal Hotel Batu Ferringhi, Penang; Malaysia; 21 November 2017 through 22 November  
2017; Code 137456

## Experimental Investigation of the Base Flow and Base Pressure of Sudden Expansion Nozzle (Conference Paper) [\(Open Access\)](#)

Saleel, A.<sup>a</sup>, Baig, M.A.A.<sup>b</sup>, Khan, S.A.<sup>c</sup>

<sup>a</sup>King Khalid University, Abha, Saudi Arabia

<sup>b</sup>School of Mechanical Engineering, CMR Technical Campus, Hyderabad, Telangana, India

<sup>c</sup>Department of Mechanical Engineering, IIUM, Kuala Lumpur, Malaysia

### Abstract

View references (27)

This paper presents an experimental investigation of an airflow from convergent-divergent axisymmetric nozzles expanded suddenly into circular duct of larger cross-sectional area than that of nozzle exit area, focusing attention on the base pressure and the flow development in the duct. To investigate the influence of area ratios and nozzle pressure ratios on the flow field developed in the duct, the micro jets of 1 mm orifice diameter located at 900 interval along a pitch circle diameter 1.3 times the nozzle exit diameter were employed as the controller of the base pressure. The Mach number investigated in the present study was 1.87, 2.2 and 2.58. The area ratios of the present study are 2.56, 3.24, 4.84 and 6.25. The nozzle pressure ratio (NPR) used were 3 and 5. The length-to-diameter ratio of the enlarged duct was varied from 10 to 1. The level of expansion at the nozzle exit (i.e. before sudden expansion) influences the wall pressure very strongly. From the results it is observed that for NPRs 3 there is no appreciable gain in the base pressure, and hence control employed in the form of micro jets is not effective for this NPR, however, at NPR 5, there is significant change in the base pressure values for all the area ratios. This clearly indicates that the level of expansion plays an important role to dictate the value of the base pressure and ultimately the control effectiveness by the micro jets. © Published under licence by IOP Publishing Ltd.

### SciVal Topic Prominence

Topic: Nozzles | Rocket nozzles | pressure ratios

Prominence percentile: 71.209

### Indexed keywords

Engineering controlled terms:

Engineering uncontrolled terms:

Engineering main heading:

Metrics View all metrics >

1 Citation in Scopus

7.81 Field-Weighted Citation Impact



### PlumX Metrics

Usage, Captures, Mentions,  
Social Media and Citations  
beyond Scopus.

### Cited by 1 document

Base pressure control by supersonic micro jets in a suddenly expanded nozzle

Khan, S.A., Chaudhary, Z.I., Shinde, V.B. (2018) *International Journal of Mechanical and Mechatronics Engineering*

[View details of this citation](#)

Inform me when this document is cited in Scopus:

[Set citation alert >](#)

[Set citation feed >](#)

### Related documents

Experimental investigation on the effectiveness of active control mechanism on base pressure at low supersonic mach numbers  
Chaudhary, Z.I., Shinde, V.B., Bashir, M.

(2017) *Lecture Notes in Mechanical Engineering*

Base pressure control using micro-jets in supersonic flow regimes

Sethuraman, V., Khan, S.A. (2018) *International Journal of Aviation, Aeronautics, and Aerospace*

## References (27)

[View in search results format >](#)

All    [Export](#)     Print     E-mail    [Save to PDF](#)    [Create bibliography](#)

- 1 Lefrançois, E., Dhatt, G., Vandromme, D.  
Fluid-structural interaction with application to rocket engines

(1999) *International Journal for Numerical Methods in Fluids*, 30 (7), pp. 865-895. Cited 23 times.  
[http://onlinelibrary.wiley.com/journal/10.1002/\(ISSN\)1097-0363](http://onlinelibrary.wiley.com/journal/10.1002/(ISSN)1097-0363)  
doi: 10.1002/(SICI)1097-0363(19990815)30:7<865::AID-FLD870>3.0.CO;2-5

[View at Publisher](#)

[View all related documents based on references](#)

Find more related documents in Scopus based on:

[Authors >](#) [Keywords >](#)

- 2 Blades, E.L., Luke, E.A., Ruf, J.  
Fully coupled fluid-structure interaction simulations of rocket engine side loads
- (2012) *48th AIAA/ASME/SAE/ASEE Joint Propulsion Conference and Exhibit 2012*. Cited 9 times.  
ISBN: 978-160086935-8

- 3 Reference information not available.

- 4 Östlund, J., Muhammad-Klingmann, B.  
Supersonic flow separation with application to rocket engine nozzles
- (2005) *Applied Mechanics Reviews*, 58 (1-6), pp. 143-176. Cited 37 times.  
doi: 10.1115/1.1894402

[View at Publisher](#)

- 5 Reference information not available.

- 6 Srivastava, N., Tkacik, P.T., Keanini, R.G.  
Influence of nozzle random side loads on launch vehicle dynamics
- (2010) *Journal of Applied Physics*, 108 (4), art. no. 044911. Cited 6 times.  
doi: 10.1063/1.3457887

[View at Publisher](#)

- 7 Reference information not available.

- 8 Östlund, J., Damgaard, T., Frey, M.  
Side-load phenomena in highly overexpanded rocket nozzles
- (2004) *Journal of Propulsion and Power*, 20 (4), pp. 695-704. Cited 53 times.

[View at Publisher](#)

Quadros, J.D. , Khan, S.A. ,  
Antony, A.J.  
(2016) *International Journal of Energy, Environment and Economics*

9 Verma, S.B., Stark, R., Haidn, O.

Relation between shock unsteadiness and the origin of side-loads inside a thrust optimized parabolic rocket nozzle

(2006) *Aerospace Science and Technology*, 10 (6), pp. 474-483. Cited 30 times.  
doi: 10.1016/j.ast.2006.06.004

[View at Publisher](#)

---

10 Reference information not available.

11 Reference information not available.

12 Menon, N., Skews, B.

3-D shock structure in underexpanded supersonic jets from elliptical and rectangular exits  
(2005) *Shock Waves: Springer*, pp. 529-534.

13 Mueller, T.J., Hall, C.R., Roache, P.J.

Influence of initial flow direction on the turbulent base pressure in supersonic axisymmetric flow

(1970) *Journal of Spacecraft and Rockets*, 7 (12), pp. 1484-1488. Cited 13 times.  
doi: 10.2514/3.30202

[View at Publisher](#)

---

14 Tanner, M.

Base cavity at angles of incidence

(1988) *AIAA Journal*, 26 (3), pp. 376-377. Cited 21 times.  
doi: 10.2514/3.9903

[View at Publisher](#)

---

15 Khan, S.A., Rathakrishnan, E.

Active control of suddenly expanded flows from overexpanded nozzles

(2002) *International Journal of Turbo and Jet Engines*, 19 (1-2), pp. 119-126. Cited 28 times.  
<http://www.degruyter.com/view/j/tjj.2012.29.issue-2/issue-files/tjj.2012.29.issue-2.xml>  
doi: 10.1515/TJJ.2002.19.1-2.119

[View at Publisher](#)

---

16 Khan, S.A., Rathakrishnan, E.

Control of Suddenly Expanded Flows with Micro-Jets

(2003) *International Journal of Turbo and Jet Engines*, 20 (1), pp. 63-81. Cited 26 times.  
<http://www.degruyter.com/view/j/tjj.2012.29.issue-2/issue-files/tjj.2012.29.issue-2.xml>  
doi: 10.1515/TJJ.2003.20.1.63

[View at Publisher](#)

17 Khan, S.A., Rathakrishnan, E.

### Control of suddenly expanded flow

(2006) *Aircraft Engineering and Aerospace Technology*, 78 (4), pp. 293-309. Cited 20 times.  
doi: 10.1108/1748840610675573

[View at Publisher](#)

---

18 Khan, S.A., Rathakrishnan, E.

### Active control of suddenly expanded flows from underexpanded nozzles

(2004) *International Journal of Turbo and Jet Engines*, 21 (4), pp. 233-253. Cited 22 times.  
<http://www.degruyter.com/view/j/tjj.2012.29.issue-2/issue-files/tjj.2012.29.issue-2.xml>  
doi: 10.1515/TJJ.2004.21.4.233

[View at Publisher](#)

---

19 Ashfaq, S., Khan, S.A., Rathakrishnan, E.

### Active Control of Flow through the Nozzles at Sonic Mach Number

(2013) *International Journal of Emerging Trends in Engineering and Development*, 2, pp. 73-82. Cited 4 times.

---

20 Chaudhary, Z.I., Shinde, V.B., Bashir, M., Khan, S.A.

### Experimental investigation on the effectiveness of active control mechanism on base pressure at low supersonic mach numbers

(2017) *Lecture Notes in Mechanical Engineering*, PartF9, pp. 197-209. Cited 3 times.  
[www.springer.com/series/11236](http://www.springer.com/series/11236)  
doi: 10.1007/978-981-10-1771-1\_24

[View at Publisher](#)

---

21 Baig, M.A.A., Al-Mufadi, F., Khan, S.A., Rathakrishnan, E.

### Control of base flows with micro jets

(2011) *International Journal of Turbo and Jet Engines*, 28 (1), pp. 59-69. Cited 12 times.  
doi: 10.1515/TJJ.2011.009

[View at Publisher](#)

---

22 Baig, M.A.A., Khan, S.A., Rathakrishnan, E.

### Active Control of Base Pressure in Suddenly Expanded Flow for Area Ratio 4.84

(2012) *International Journal of Engineering Science and Technology*, 4, pp. 1892-1902. Cited 4 times.

---

23 Reference information not available.

---

24 Chaudhary, Z., Shinde, V.B., Bashir, M., Khan, S.

Experimental investigation of the base flow from the nozzles with sudden expansion  
(2015) *International Journal of Applied Engineering Research*, 10.

---

25 Reference information not available.

- 26 Baig, M.A.A., Khan, S.A., Rathakrishnan, E.  
Control of Nozzle Flow in Suddenly Expanded Duct with Micro Jets  
*International Journal of Engineering Science & Advanced Technology [IJESAT]*, 2, pp. 789-795.

- 
- 27 Ashfaq, S., Khan, S., Rathakrishnan, E.  
Control of suddenly expanded flow for area ratio 3.61  
*International Journal of Advanced Scientific and Technical Research*, 2013, pp. 798-807.
- 

© Copyright 2018 Elsevier B.V., All rights reserved.

< Back to results | 1 of 12 Next >

^ Top of page

#### About Scopus

[What is Scopus](#)  
[Content coverage](#)  
[Scopus blog](#)  
[Scopus API](#)  
[Privacy matters](#)

#### Language

[日本語に切り替える](#)  
[切换到简体中文](#)  
[切换到繁體中文](#)  
[Русский язык](#)

#### Customer Service

[Help](#)  
[Contact us](#)

**ELSEVIER**

[Terms and conditions](#) ↗ [Privacy policy](#) ↗

Copyright © 2019 Elsevier B.V. All rights reserved. Scopus® is a registered trademark of Elsevier B.V.  
We use cookies to help provide and enhance our service and tailor content. By continuing, you agree to the  
use of cookies.

 RELX Group™