SELECTED TOPICS

Im Aerospace Engineering

EDITOR

ERWIN SULAEMAN





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SELECTED TOPICS In Aerospace Engineering

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TABLE OF CONTENT

PR	EFACE		
TA	BLE OF	F CONTENT	ii
CO	NTRIB	UTING AUTHORS	xii
PA.	RT I		
FU	NDAME	ENTAL COMPRESSOR FOR AIRCRAFT'S TURBO	
EN	GINES		1
NO	MENCL	ATURE.	2
СН	APTER	ONE	
IN	TRODUC	CTION TO COMPRESSORS.	4
1.1	Introduc	tion	4
1.2	Types of	Compressors	4
	1.2.1	Positive Displacement Compressors	4
	1.2.2	Dynamic Compressors	4
1.3	3 Comparison of Compressor Types		5
	1.31	Flow rate	5
	1.3.2	Efficiency	6
	1.3.3	Pressure Ratio	7
	1.3.4	Characteristic Curves	7

CHAPTER TWO

TW	O-DIMENSIONAL ANALYSIS OF COMPRESSORS.	9
2.1	Velocity diagrams of the compressor stage	10
2.2	Thermodynamics of the compressor stage	10
2.3	Stage loss and efficiency	11
2.4	Reaction ratio	12
2.5	Stage loading	13
СН	APTER THREE	
TH	REE-DIMENSIONAL ANALYSIS OF COMPRESSORS.	15
3.1	Theory of radial equilibrium	15
3.2	Free-vortex flow	16
3.3	Forced vortex	18
3.4	General whirl distribution	18
СН	APTER FOUR	
RO	TATING STALL AND SURGE.	20
4.1	Performance of Axial and Radial Compressors	20
4.2	Aerodynamic Flow Instabilities	22
	4.2.1 Rotating stall	23
	4.2.2 Surge	24
	4.2.3 Rotating Stall and Surge in Radial Compressors	26
СН	APTER FIVE	
MC	DDELING OF COMPRESSION SYSTEMS.	27
5.1	Introduction	27
5.2	Greitzer lumped parameter model	28
СН	APTER SIX	
CO	MPRESSOR MODELS.	34

6.1	Moore Model	34
6.2	Moore-Greitzer model	36
CH	IAPTER SEVEN	
SU.	RGE AND ROTATING STALL.	40
7.1	Stability of compression systems	40
7.2	Control of Surge and Rotating Stall	41
7.3	Avoidance Control	41
7.4	Active Control	44
RE	FERENCE OF PART I.	46
PA.	RT II	
RIC	GID-BODY DYNAMICS OF AIR VEHICLE	48
СН	IAPTER EIGHT	
AII	RCRAFT RIGID-BODY EQUATION OF MOTIONS: A	
NO	NLINEAR MODEL	
8.1	Introduction	49
8.2	Definition of Axes and Angles	49
8.3	The Rigid-Body Equations	52
8.4	Conclusions	55
СН	APTER NINE	
AII	RCRAFT EQUATIONS OF MOTIONS: A NONLINEAR	R MODEL
9.1	Introduction	57
9.2	Orientation and Position of the Airplane	57
9.3	Euler's Equations of Motion	59
9.4	Effect of Spinning Rotors	60
9.5	The Collected Equations	61

9.6 Conclusions	62
CHAPTER TEN	
AIRCRAFT EQUATIONS OF MOTION: A LINEAR MODEL	
10.1 Introduction	63
10.2 The Small-Disturbance Theory	63
10.3 Conclusions	69
REFERENCE OF PART II	70
PART III	
DYNAMICS OF FLEXIBLE STRUCTURE OF AIR	
VEHICLE	71
NOMENCLATURES	72
CIVA DEED EL EVEN	
CHAPTER ELEVEN	
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE	76
	76 76
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE	
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction	76 76
 OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 	76
 OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 	76 76 77 79
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 11.4 Aerodynamic Discrete Element Methods 11.5 The doublet lattice method (DLM) 11.6 The doublet point method (DPM)	76 76 77 79 79 80
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 11.4 Aerodynamic Discrete Element Methods 11.5 The doublet lattice method (DLM)	76 76 77 79
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 11.4 Aerodynamic Discrete Element Methods 11.5 The doublet lattice method (DLM) 11.6 The doublet point method (DPM)	76 76 77 79 79 80
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 11.4 Aerodynamic Discrete Element Methods 11.5 The doublet lattice method (DLM) 11.6 The doublet point method (DPM) 11.7 Conclusions	76 76 77 79 79 80
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 11.4 Aerodynamic Discrete Element Methods 11.5 The doublet lattice method (DLM) 11.6 The doublet point method (DPM) 11.7 Conclusions CHAPTER TWELVE	76 76 77 79 79 80
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 11.4 Aerodynamic Discrete Element Methods 11.5 The doublet lattice method (DLM) 11.6 The doublet point method (DPM) 11.7 Conclusions CHAPTER TWELVE TRANSLATION OF AXIS PROCEDURE TO CONSTRUCT	76 76 77 79 79 80 81
OVERVIEW OF DYNAMICS OF FLEXIBLE AIR VEHICLE 11.1 Introduction 11.2 The Influence of the Structural Flexibility on Vehicle Design 11.3 Non-uniform Beam Finite Element 11.4 Aerodynamic Discrete Element Methods 11.5 The doublet lattice method (DLM) 11.6 The doublet point method (DPM) 11.7 Conclusions CHAPTER TWELVE TRANSLATION OF AXIS PROCEDURE TO CONSTRUCT STIFFNESS MATRIX	76 76 77 79 79

12.4 Stiffness Matrix Construction	85
12.5 Conclusion	87
CHAPTER THIRTEEN	
MINIMUM DENOMINATOR OF RATIONAL FUNCTION	
13.1 Introduction	88
13.2 Rational Function Transformation	88
13.3 MDRF Procedure for Non-linear Variation of the Stiffness Distribution	89
13.4 Direct Differentiation Method	90
13.5 Substitution Procedure	91
13.6 Conclusion	93
CHAPTER FOURTEEN	
	174
TORSIONAL STIFFNESS MATRIX OF NON-PRISMATIC BEA	1 /VI
ELEMENTS	
14.1 Introduction	94
14.2 Torsional - Twist Deformation Relation	94
14.3 Deformation of the Cantilever Bar Problem	95
14.4 Flexibility Matrix of the Cantilever Bar	97
14.5 Stiffness Matrix	97
14.6 Conclusion	98
CHAPTER FIFTEEN	
BENDING STIFFNESS MATRIX OF NON-PRISMATIC BEAM	
ELEMENTS	
15.1 Introduction	99
15.2 Load - Displacement Relation	99
15.3 Displacement of a Cantilever Bar Problem	100
15.4 Flexibility Matrix of the Cantilever Beam	103
15.5 Stiffness Matrix	104
15.6 Conclusion	104

CHAPTER SIXTEEN

FORMULATION OF KERNEL FUNCTION FOR AERODYNAMIC LOADING ON AIR VEHICLE

16.1 Introduction	105
16.2 Formulations of the Kernel Function	105
16.3 The formulation of Watkins, Runyan and Woolston	106
16.4 Formulations of Laschka	107
16.5 Formulations of Yates	109
16.6 Formulations of Landahl	109
16.7 Conclusion	110
CHAPTER SEVENTEEN	
UNSTEADY AERODYNAMIC THEORY OF LIFTING SUR	FACE
17.1 Introduction	111
17.2 Assumptions	111
17.3 Basic Concept	111
17.4 Boundary Conditions	113
17.5 Kernel Function	113
17.6 Incomplete Cylindrical Function	115
17.7 Conclusion	115
CHAPTER EIGHTEEN	
NUMERICAL EVALUATIONS OF HYPERGEOMETRIC	
CYLINDRICAL FUNCTIONS	
18.1 Introduction	117
18.2 Kernel Integral Function	117
18.3 Modified Bessel Function of the First Kind of Order 0	118
18.4 Modified Bessel Function of the First Kind of Order 1	119
18.5 Modified Bessel Function of the Second Kind of Order 0	120
18.6 Modified Bessel Function of the Second Kind of Order 1	121
18.7 Modified Struve Function	121
18.8 Conclusion	122

CHAPTER NINETEEN	
ANALYTICAL DERIVATION OF THE INCOMPLETE	
CYLINDRICAL FUNCTIONS: REAL PARTS	
19.1 Introduction	123
19.2 The finite subinterval of the integral	123
19.3 The Infinite Subinterval of the Integral	125
19.4 Conclusion	129
CHAPTER TWENTY	
ANALYTICAL DERIVATION OF THE INCOMPLETE	
CYLINDRICAL FUNCTIONS: IMAGINARY PARTS	
20.1 Introduction	130
20.2 The finite subinterval of the integral	130
20.3 The Infinite Subinterval of the Integral	132
20.4 Conclusion	134
CHAPTER TWENTY ONE	
ALTERNATE EXPANSION SERIES FOR THE INCOMPLETE	
CYLINDRICAL FUNCTION	
21.1 Introduction	135
21.2 Separation of Real and Imaginary Functions	135
21.3 Separation of Regular and Singular Functions	138
21.4 Conclusion	139
CHAPTER TWENTY TWO	
EXPANSION SERIES OF CONTINUOUS FUNCTION USING	
ANALYTICAL INTEGRATION OF	
LEAST SQUARE REGRESSION	
22.1 Introduction	140

22.2 Taylor and Maclaurin expansion series	140
22.3 Present Least Square Expansion Series	141
22.4 Application of the Present Approach to the Incomplete Cylindrical Function	143
22.4 Conclusion	145
CHAPTER TWENTY THREE	
ALTERNATE APPROXIMATE FUNCTION FOR KERNEL	
FUNCTION OF PLANAR OSCILLATING LIFTING SURFACES	
23.1 Introduction	146
23.2 Epstein's Approach	146
23.3 Present Approach for Near Field Region	147
23.4 Present Approach for Middle Field Region	150
23.5 Present Approach for Far Field Region	151
23.6 Conclusion	151
CHAPTER TWENTY FOUR	
APPROXIMATE FUNCTION FOR NEAR-FIELD KERNEL	
FUNCTION OF NON-PLANAR LIFTING SURFACES	
24.1 Introduction	152
24.2 Kernel Function Equation	152
24.3 Present Approximate Function	154
24.4 Conclusion	157
CHAPTER TWENTY FIVE	
APPROXIMATE FUNCTION FOR FAR-FIELD KERNEL	
FUNCTION OF OSCILLATING NON-PLANAR LIFTING	
SURFACES	
25.1 Introduction	158
25.2 Landahl's Kernel Function Equation	158
25.3 Present Kernel Function Formulation	159
25.4 Conclusion	161

IMPROVED VORTEX LATTICE METHOD	
26.1 Introduction	162
26.2 Present Vortex Lattice Method	162
26.3 Conclusion	167
CHAPTER TWENTY SEVEN	
IMPROVED DOUBLET POINT METHOD	
27.1 Introduction	168
27.2 Present DPM for Planar Lifting Surfaces	168
27.3 Present DPM for Non-Planar Lifting Surfaces	170
27.4 Conclusion	174
CHAPTER TWENTY EIGHT	
IMPROVED DOUBLET LATTICE METHOD	
28.1 Introduction	176
28.2 Present DLM for Planar Lifting Surfaces	176
28.3 Conclusion	179
CHAPTER TWENTY NINE	
APPLICATION OF THE AERODYNAMIC DISCRETE ELEM	1ENT
METHODS	
29.1 Introduction	180
29.2 Delta Wing with AR=2	180
29.3 Cropped-Double-Delta Wing	182
29.4 Sweptback Wing with Partial Flap	183
29.5 AGARD Wing-Horizontal Tail	184
29.6 Conclusion	186
CHAPTER THIRTY	
AEROELASTIC STABILITY PROBLEM OF AIR VEHICLE	
30.1 Introduction	187

CHAPTER TWENTY SIX

REFERENCES OF PART III	195
30.4 Conclusion	193
30.3 Validation of the present flutter procedure	191
30.2 The Flutter Solution Method	187

SURGE AND ROTATING STALL

7.1 Stability of compression systems

stability analysis can be performed based on the eigenvalues of the linearized, second order Greitzer lumped parameter model. At this point, we introduce the characteristic equation (Greitzer, 1981):

$$s^{2} + \left[\frac{1}{B \frac{d\Psi_{t}}{d\Phi_{t}}} - B \frac{d\Psi_{c}}{d\Phi_{c}} \right] s + \left[1 - \frac{\frac{d\Psi_{c}}{d\Phi_{c}}}{\frac{d\Psi_{t}}{d\Phi_{t}}} \right] = 0$$
 (7.1)

where is $\frac{d\Psi_1}{d\Phi_1}$ the slope of the dimensionless throttle characteristic and $\frac{d\Psi_c}{d\Phi_c}$ is the slope of the dimensionless compressor characteristic. Thus, the linearized compression system is equivalent to a damped linear oscillator:

$$s^2 + 2\varsigma \omega_0 s + \omega_0^2 = 0 (7.2)$$

Application of the Routh-Hurwitz criterion gives that the system is strictly Hurwitz, *i.e.*, all the roots have negative real parts, if and only if,

$$\varsigma \ \omega_0 > 0 \quad and \quad \omega_0^2 > 0 \tag{7.3}$$

Comparison of the previous two equations learns that both stability conditions for the compression system depend on the slopes of the dimensionless compressor and throttle characteristics,

$$\frac{d\Psi_{c}}{d\Phi_{c}} < \frac{d\Psi_{t}}{d\Phi_{t}} \quad and \quad \frac{d\Psi_{c}}{d\Phi_{c}} < \frac{1}{B^{2} \frac{d\Psi_{t}}{d\Phi_{t}}}$$
(7.4)